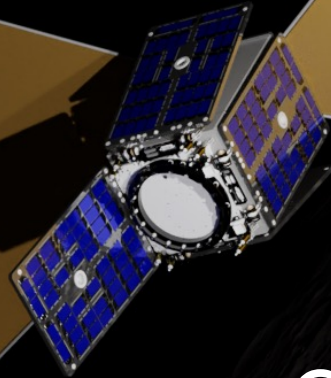


Eloy de Lera Acedo
University of Cambridge, UK



CosmoCube

Probing the Cosmic Dark Ages
with a Miniature Radiometer in Lunar Orbit

09/07/2025

NAM 2025, Durham, UK

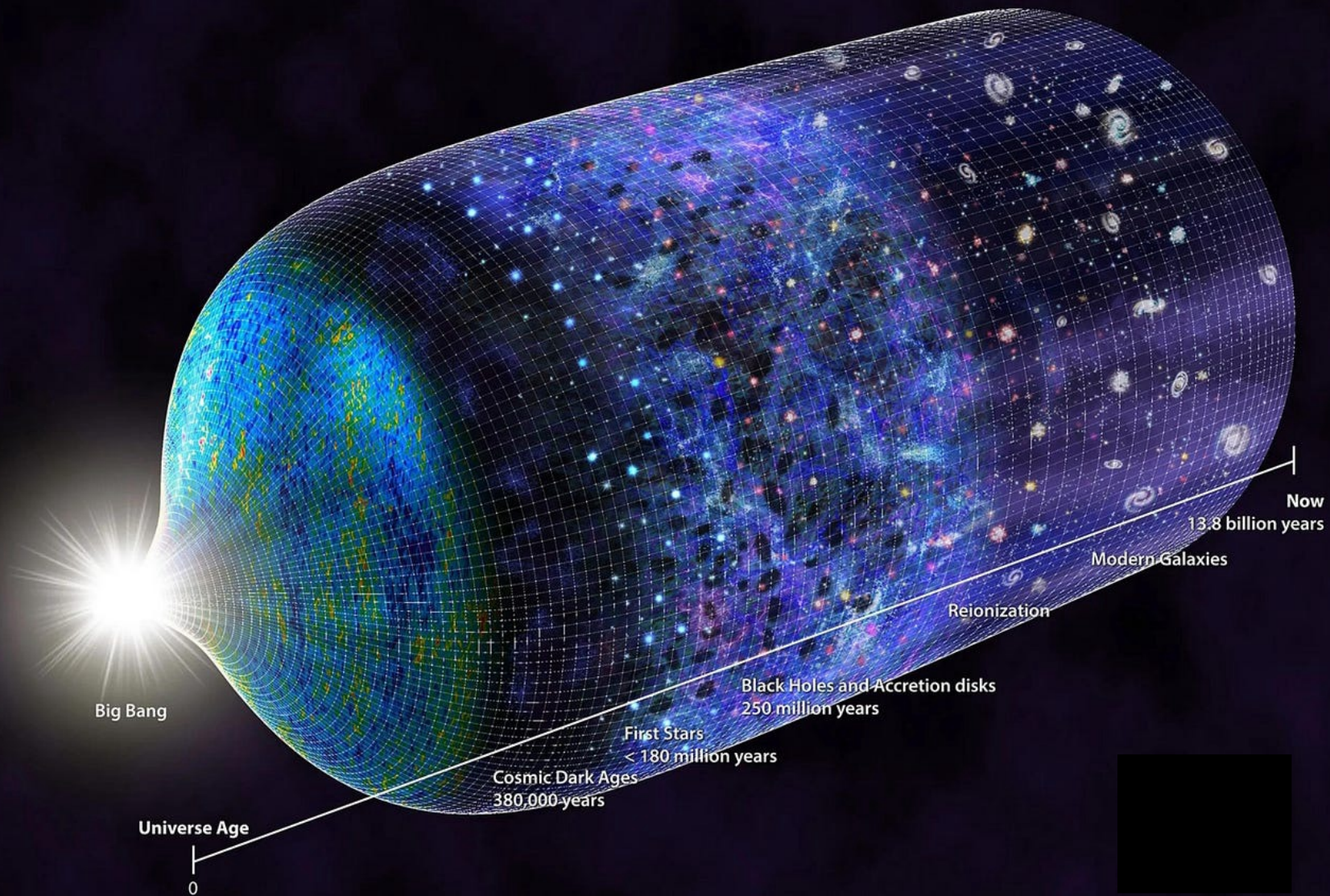


Image credit: NSF

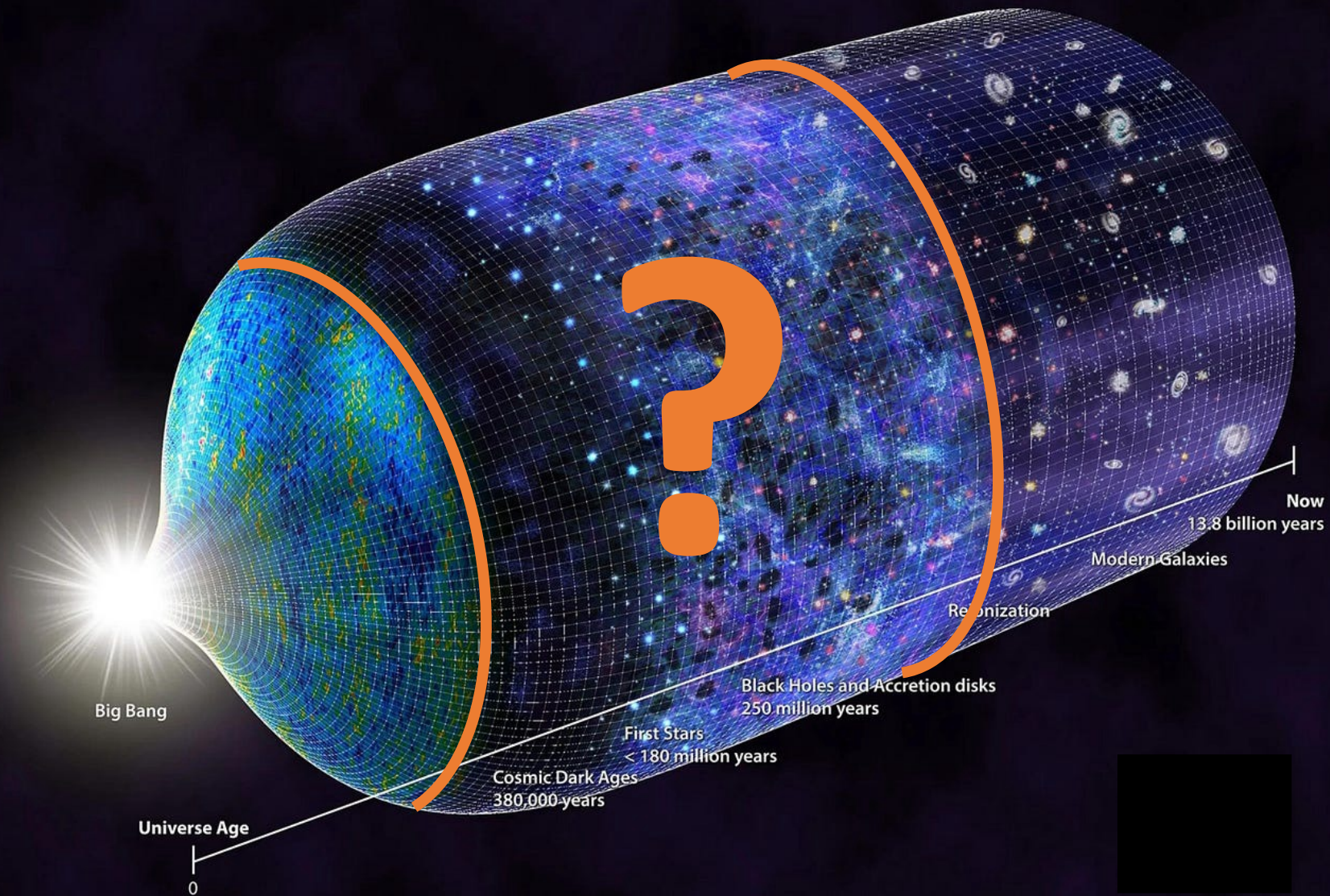


Image credit: NSF

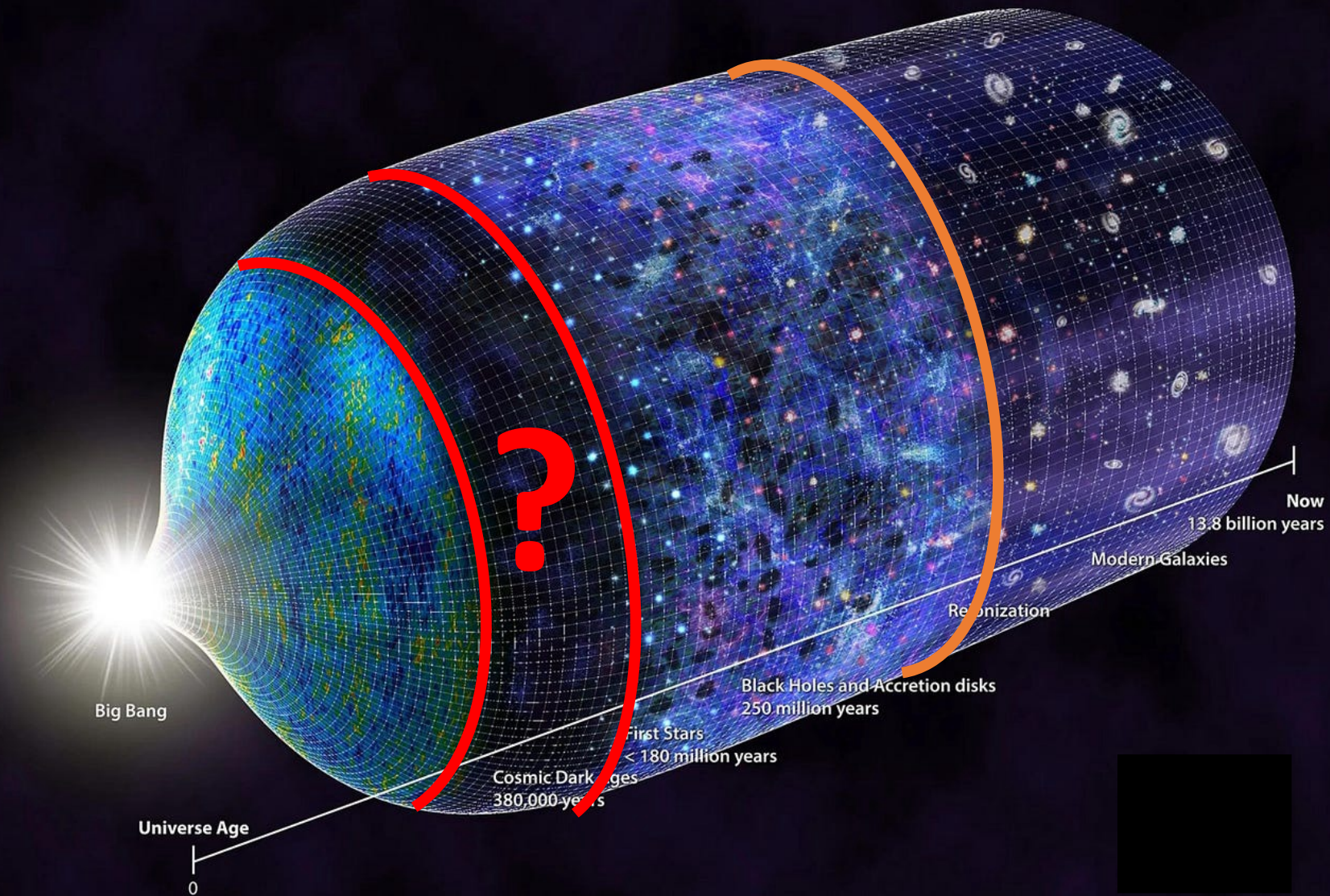
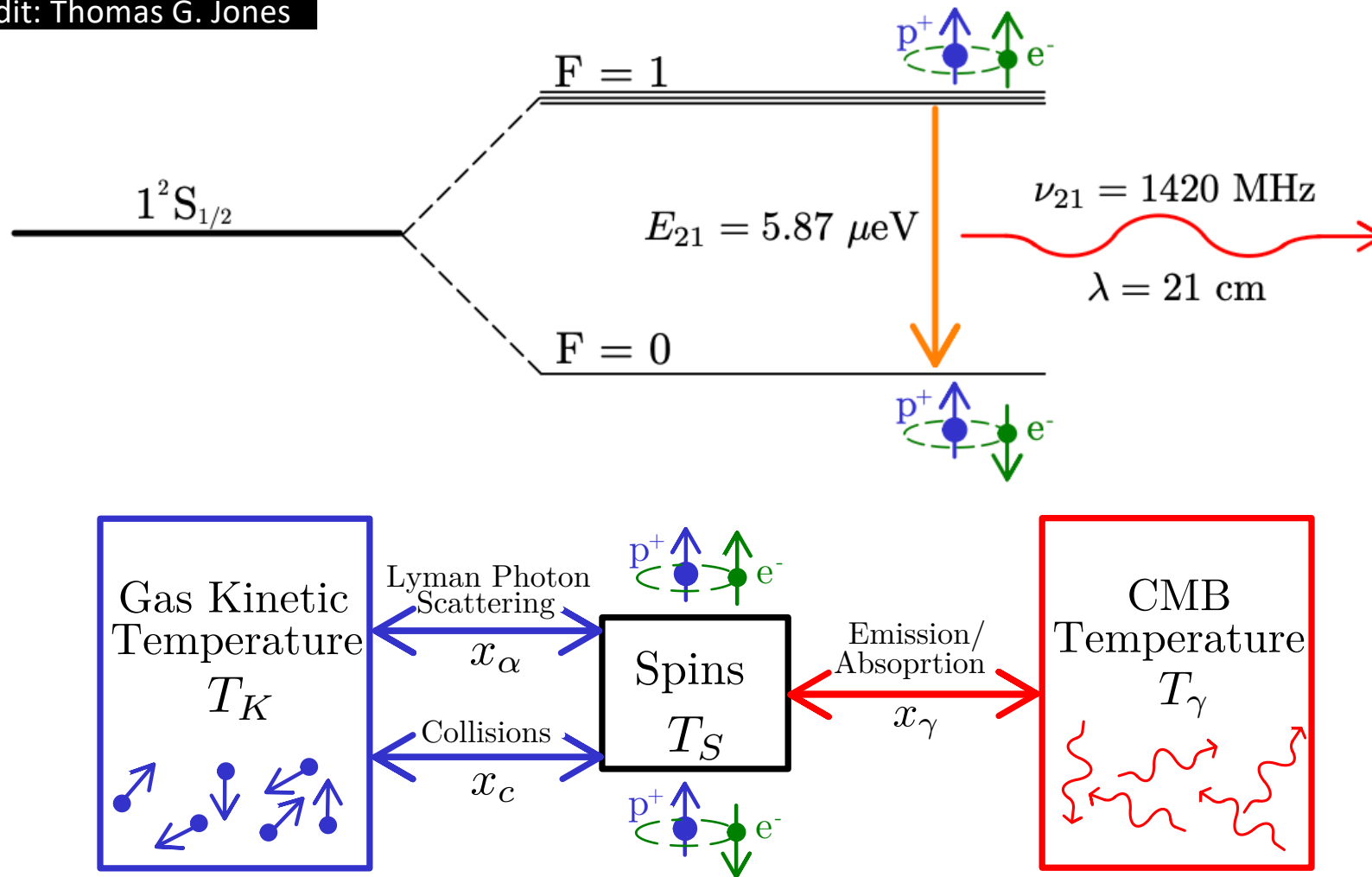
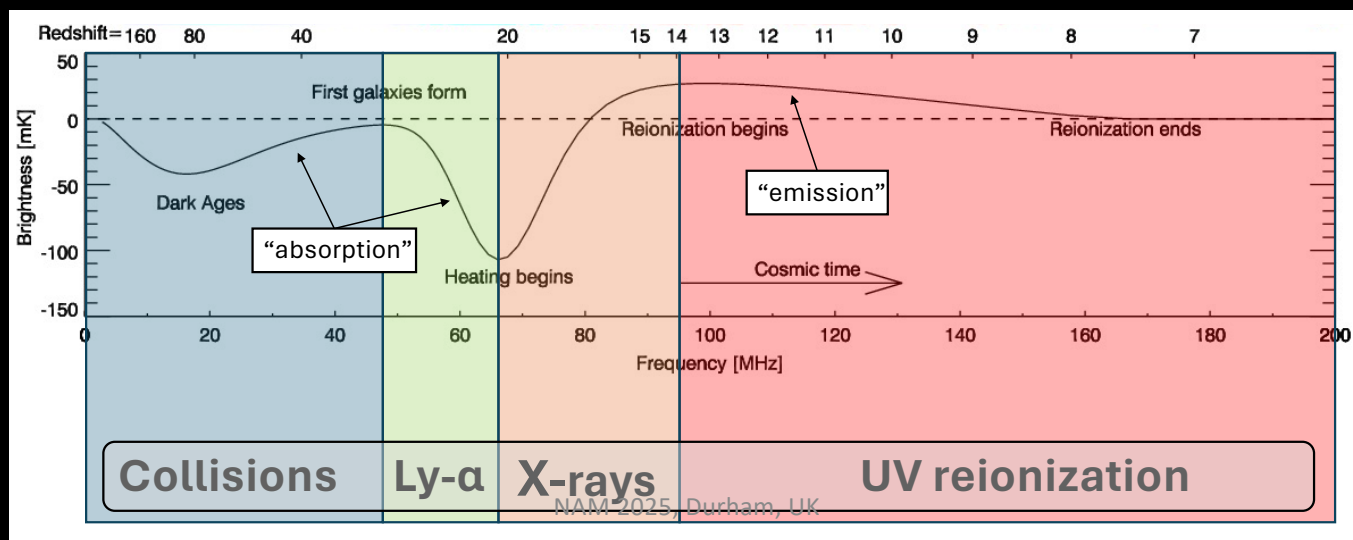
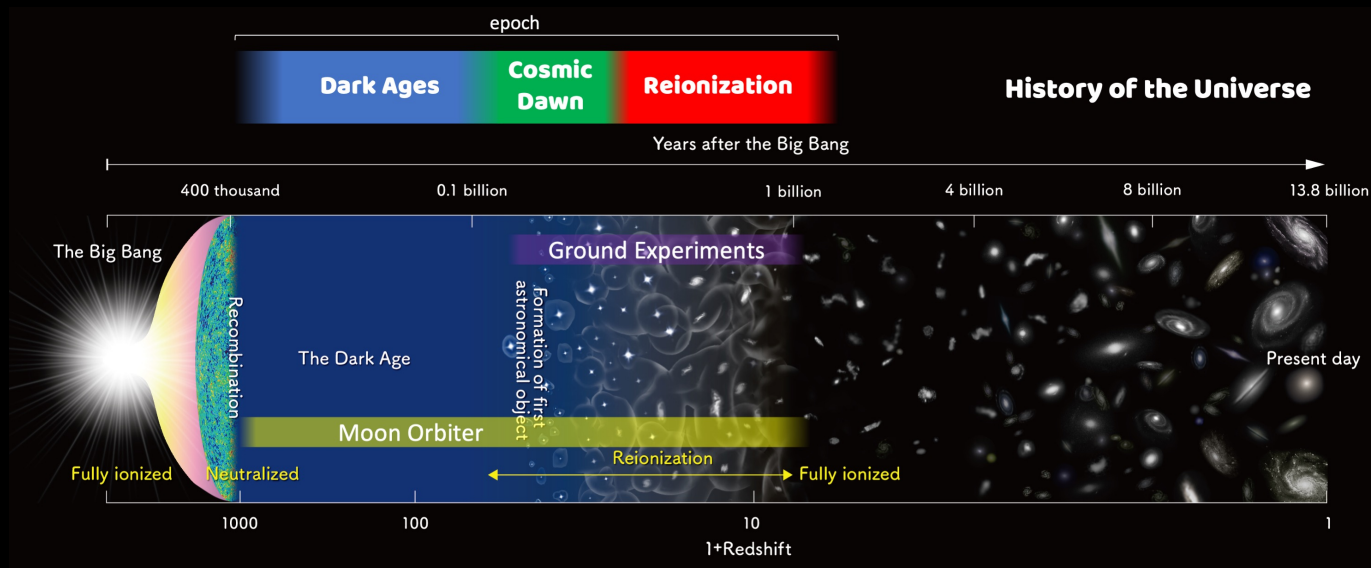


Image credit: NSF

Image credit: Thomas G. Jones





Ground-based experiments



REACH



EDGES



SARAS



PRIZM

NAM 2025, Durham, UK

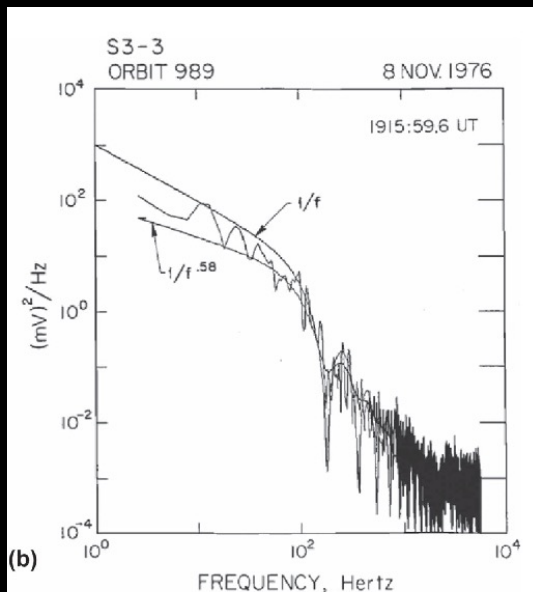
Foreground power: $\sim 5,000.0$ K

Systematics: ~ 100.0 K

21-cm signal: ~ 0.1 K

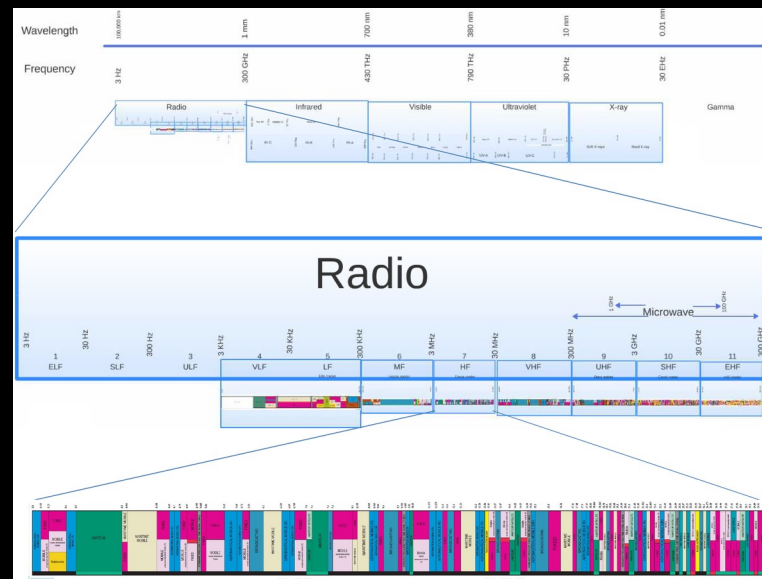
Exquisite foreground and radiometer calibration required

Challenges on the ground



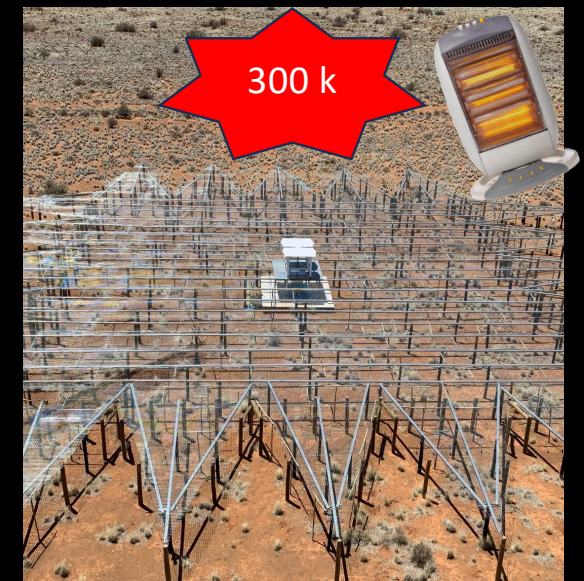
Earth's Ionosphere

09/07/2025



RFI

NAM 2025, Durham, UK



Near field
emissions

Solutions in space

Earth's Ionosphere

RFI

Near field emissions

A **space mission** on
the **far side of the**
Moon, in **orbit**

Unique opportunity
if humankind's
history, first time we
are capable of doing
this, but maybe the
last time too...

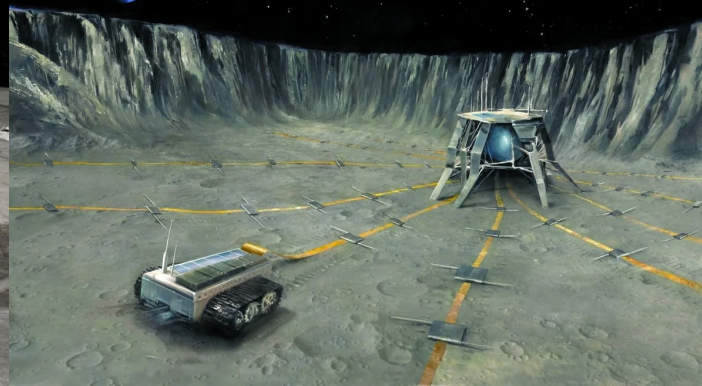
The Moon is back on the agenda



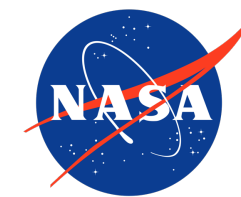
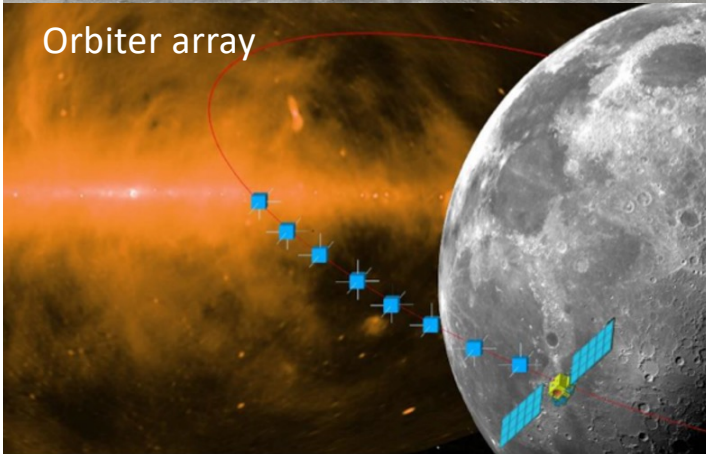
Single-antenna lander



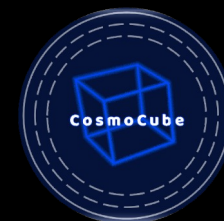
Surface array



Orbiter array



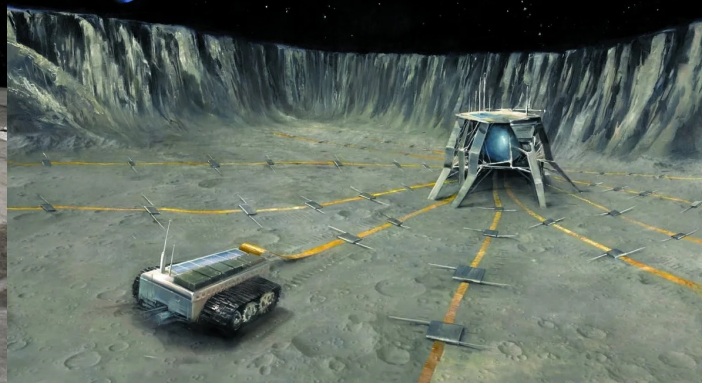
The Moon is back on the agenda



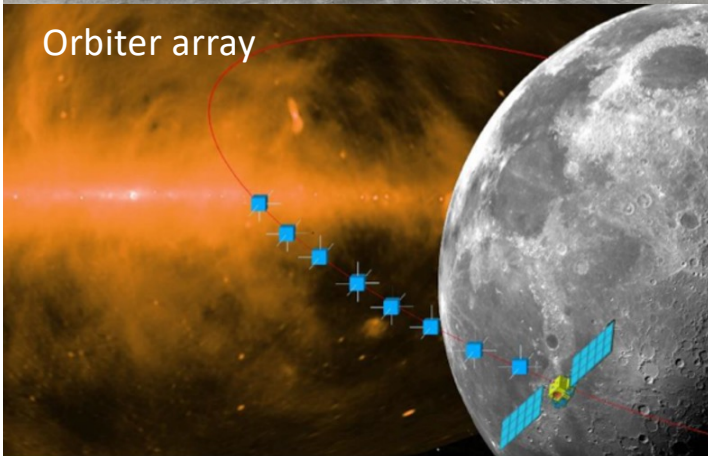
Single-antenna lander



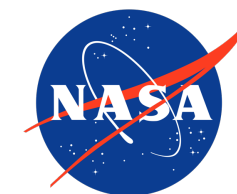
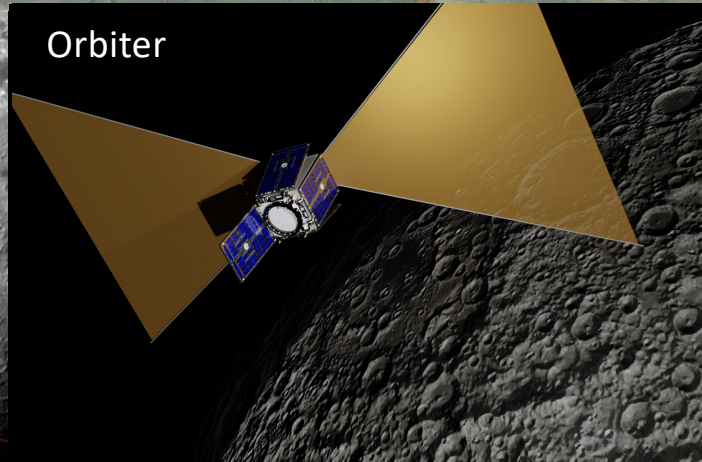
Surface array



Orbiter array



Orbiter



Funding



Core members



UNIVERSITY OF
CAMBRIDGE



UNIVERSITY OF
PORTSMOUTH



Collaborators

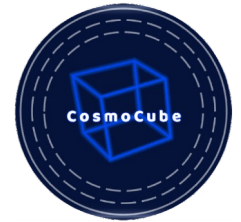


Jet Propulsion Laboratory
California Institute of Technology

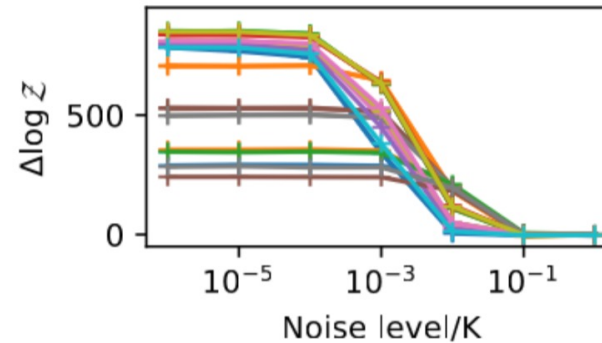
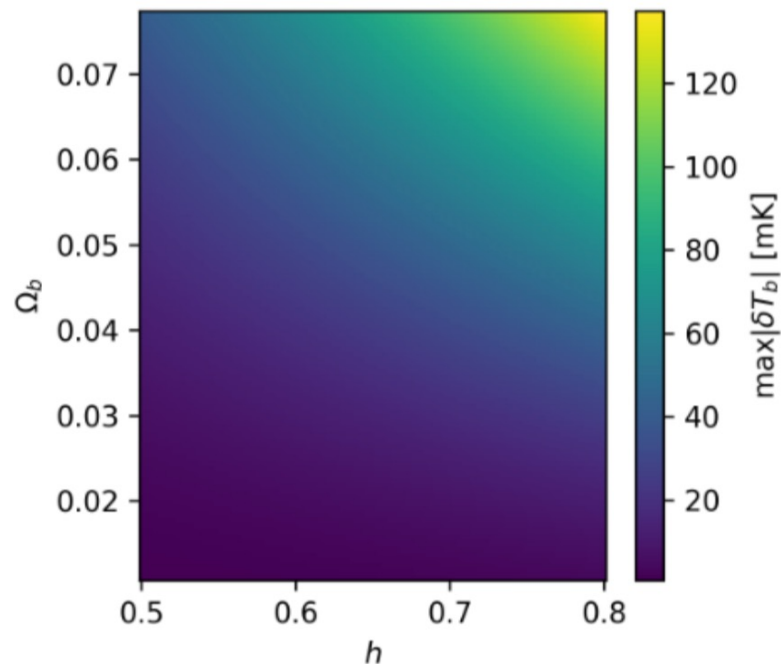


**L-Università
ta' Malta**

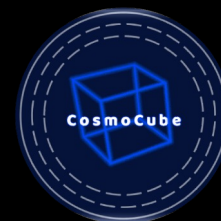




Requirements

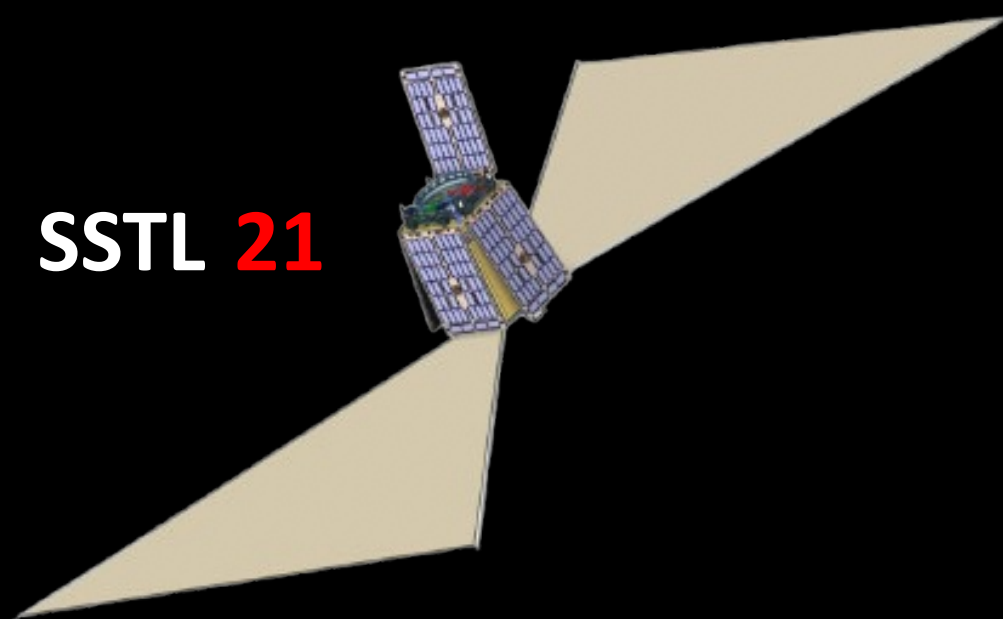


CosmoCube will be sensitive
to baryon over density –
hubble constant fluctuations
with a radiometer noise of **~10**
mK across 8-45 MHz in a 2-
year mission in Moon's orbit



The platform and payload

SSTL 21



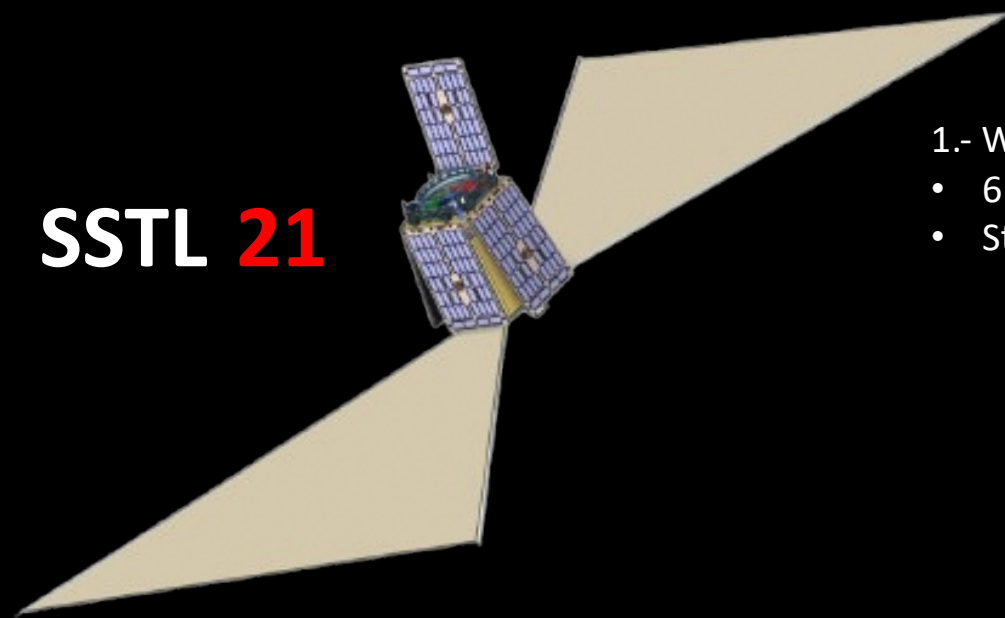
* Kaan and de Lera Acedo, RASTI 2024

~ Zhu, de Lera Acedo et al., RASTI 2024



The platform and payload

SSTL 21



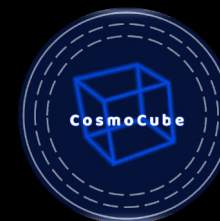
1.- Wideband deployable antenna

- 6 m long once deployed
- Stowed during launch



* Kaan and de Lera Acedo, RASTI 2024

~ Zhu, de Lera Acedo et al., RASTI 2024



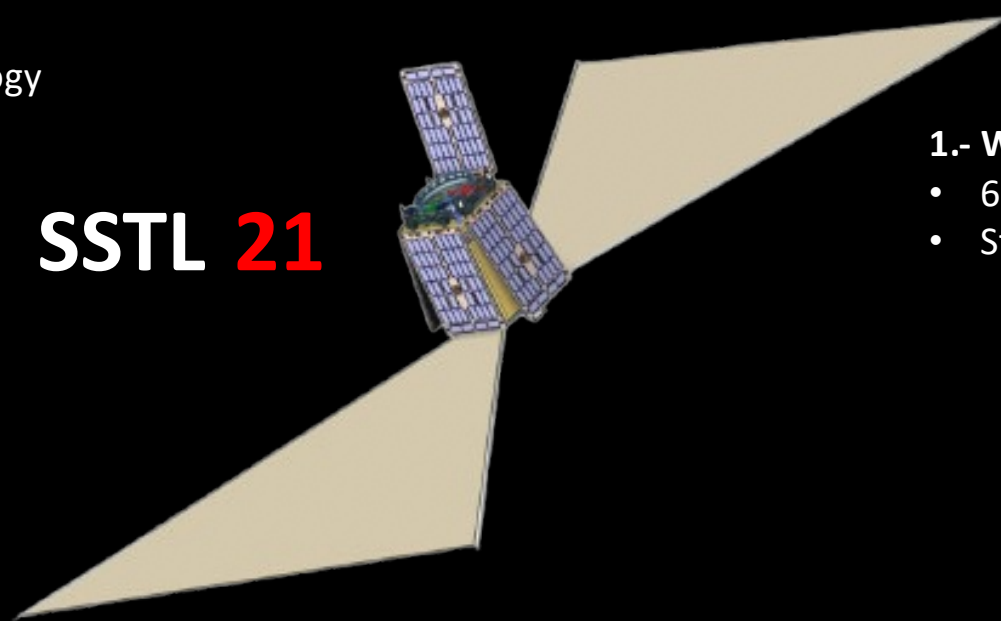
The platform and payload

2.- Integrated radiometer

- Using COTS RFSoc technology
- Dicke-switch calibrator



SSTL 21



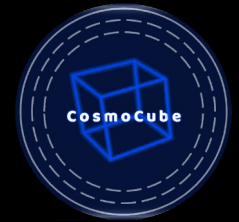
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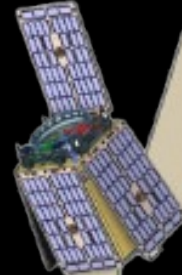
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SSTL 21

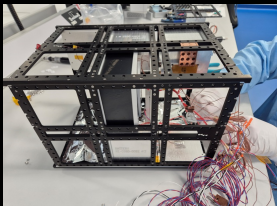


1.- Wideband deployable antenna

- 6 m long once deployed
- Stowed during launch

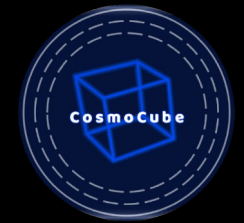
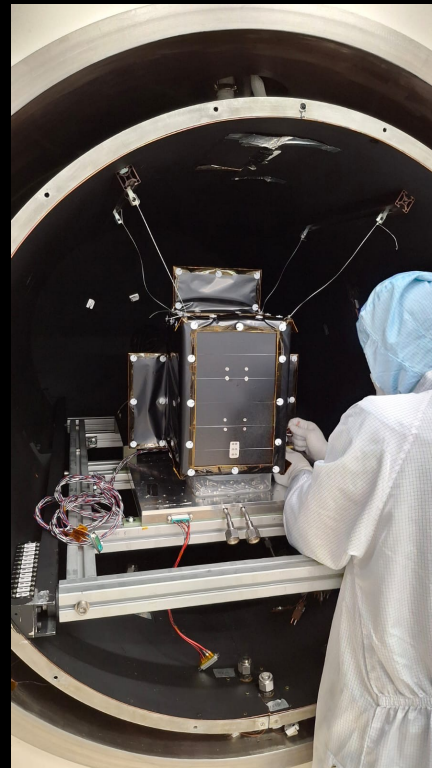
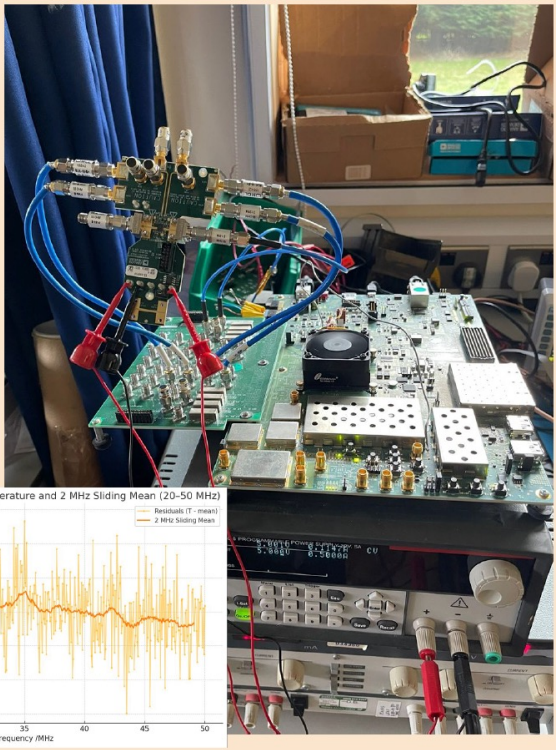
3.- Thermal control

- 10 mK rms during observations



Mission status

Full system

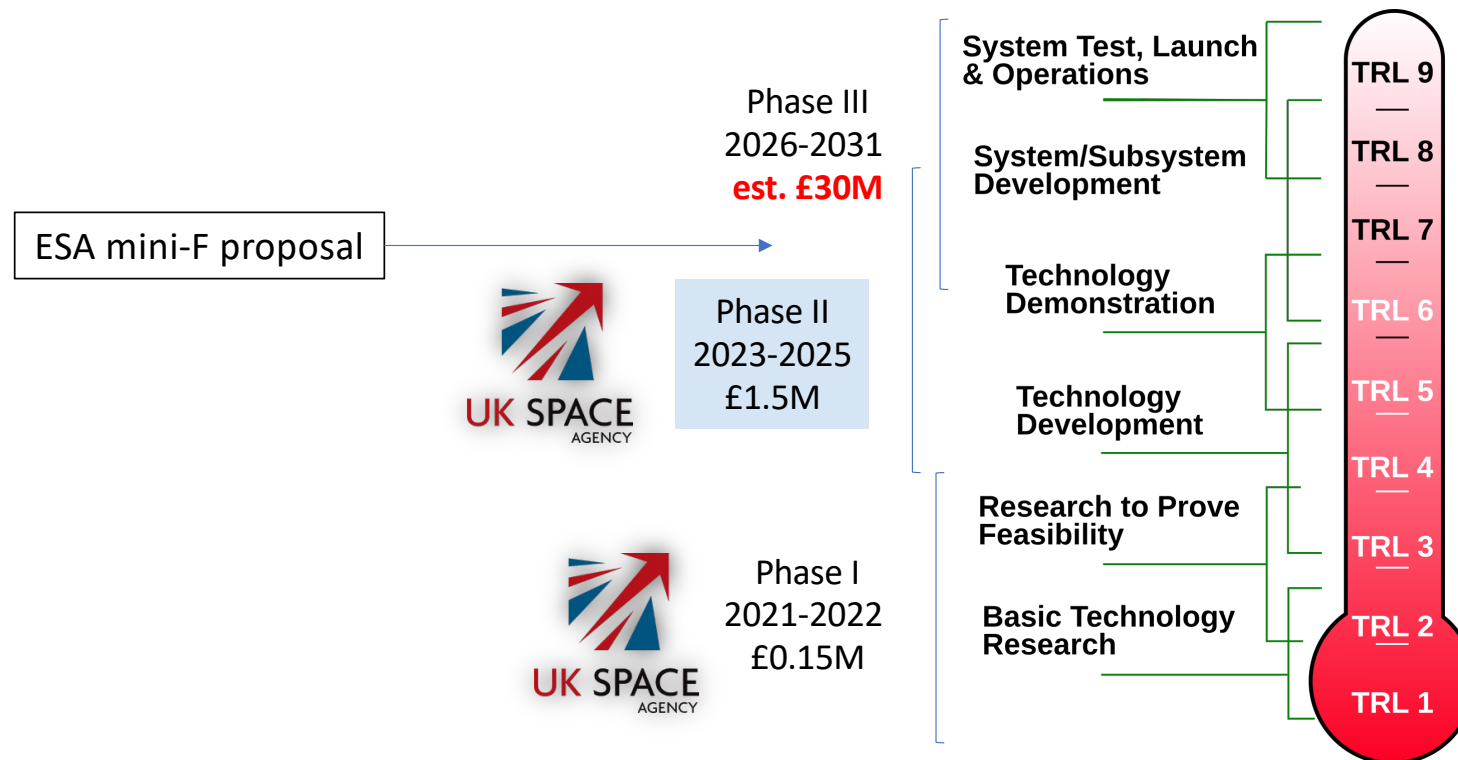


09/07/2025

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Mission status



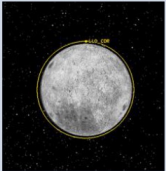
www.cosmocube.net



NAM 2025, Durham, UK

The orbit



	Circular polar orbit	LLO 100	Frozen orbit 41x190	Frozen orbit 70x166	Frozen orbit 100x199	Frozen orbit 90x190
						
Hp x Ha (km x km)	100 x 100	100.58 x 100.62	41 x 190	70.537 x 166.663	100x199	90x190
a (km)	1837.4	1838.00	1853.201	1856	1887.117	1877.3639
e	0	1.00E-05	0.04036	0.025896	0.026291	0.0266139
i (deg)	90	94	90	91	90	90
RAAN (deg)	0	0	0	0	0	0
AoP (deg)	270	90	270	270	270	270
M (deg)	0	0	0	0	0	0
Time in RF shield per year [hr]	998	1023.7	957	955	840	872
Time in RF shield and Sun eclipse per year [hr]	389	415.72	363	373	307	324
Minimum lifetime for 1296 in RF shield [years]	1.30	1.27	1.35	1.36	1.54	1.49
Minimum lifetime for 1296 with in RF shield and Sun eclipse [years]	3.33	3.11	NAM 2025, Durham, UK 3.57	3.47	4.22	4

09/07/21

Dealing with the Sun



Reducing the impact
of the **Earth** and the
Sun

